



# Mirror Technology Roadmap for NASA's Exoplanet Exploration Program

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with

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Jet Propulsion Laboratory
California Institute of Technology

Mirror Technology SBIR/STTR Workshop, Greenbelt, Maryland, USA 21 June 2011

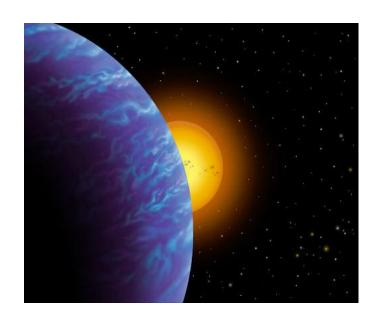
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## **Exoplanet Exploration Program Science Goals**



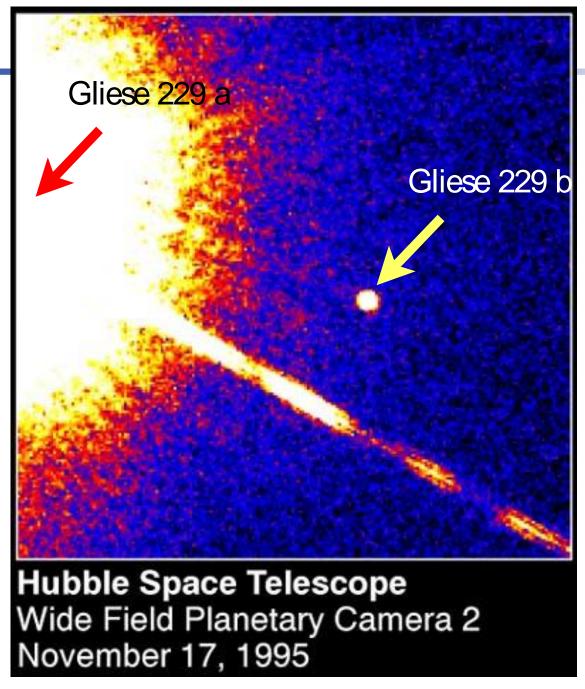
- Direct detection of terrestrial planets in the habitable zone around nearby stars
- Characterization of planetary atmospheres in search of the signatures of life
- Direct detection and characterization of other constituents of planetary systems
- Revolutionary general astrophysics investigations



- Direct detection must separate planet light from star light
- Planet characterization must determine type of planet, its gross physical properties and atmosphere constituents allowing assessment of likelihood of life







**Brown Dwarf** 44 AU 20-50 M<sub>J</sub>

Discovered 1995

Starlight suppression allows the spectra of planets to be detected

2.4-m telescope 50 mas resolution





At mid-infrared wavelengths, exoplanets shine because they are warm

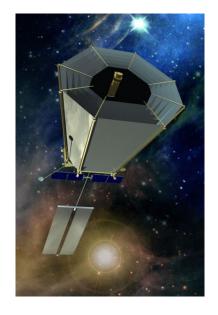
At visible wavelengths, exoplanets shine in reflected starlight

 $\wedge (\mu m)$ 



## Gallery of Exoplanet Mission Concepts

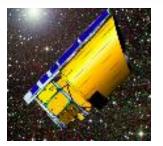




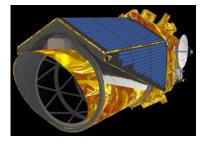
**TPF-C** Terrestrial Planet Finder Coronagraph



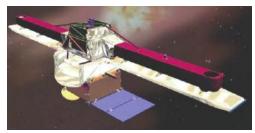
**TPF-I** Terrestrial Planet Finder Interferometer



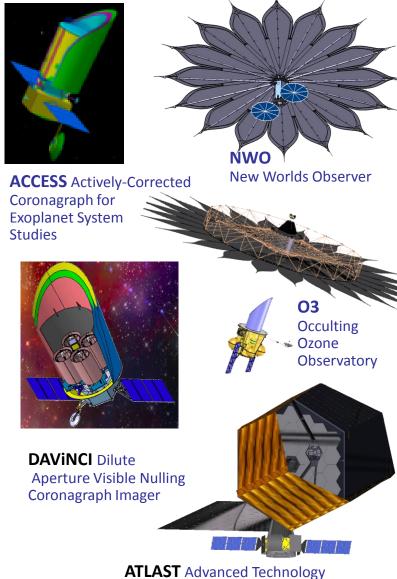
**PECO** Pupil-Mapping Exoplanet Coronagraphic Observe



**EPIC** Extrasolar Planetary Imaging Coronagraph



**FKSI** Fourrier-Kelvin Stellar Interferometer



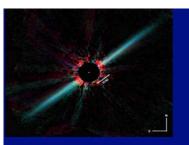
Large-Aperture Space Telescope



## Astrophysics 2010 Decadal Survey



#1 Priority for Medium-scale space program



# New Worlds Technology Development Program

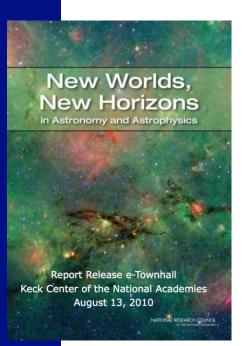
- To achieve New Worlds objective studying nearby, habitable exoplanets need preliminary observations before choosing a flagship mission:
  - Planetary demography over wide range of conditions:
    - Kepler, WFIRST, integrated ground-based program
  - Measurement of zodiacal light:
    - Ground-based telescopes.
    - Sub-orbital and explorer mission opportunities.
- In parallel, need technology development for competing approaches to make informed choice in second half of decade
- RECOMMEND \$100-200M over decade
- Planned integrated ground-space exoplanet program

New Worlds, New Horizons in Astronomy and Astrophysics

21 June 2011

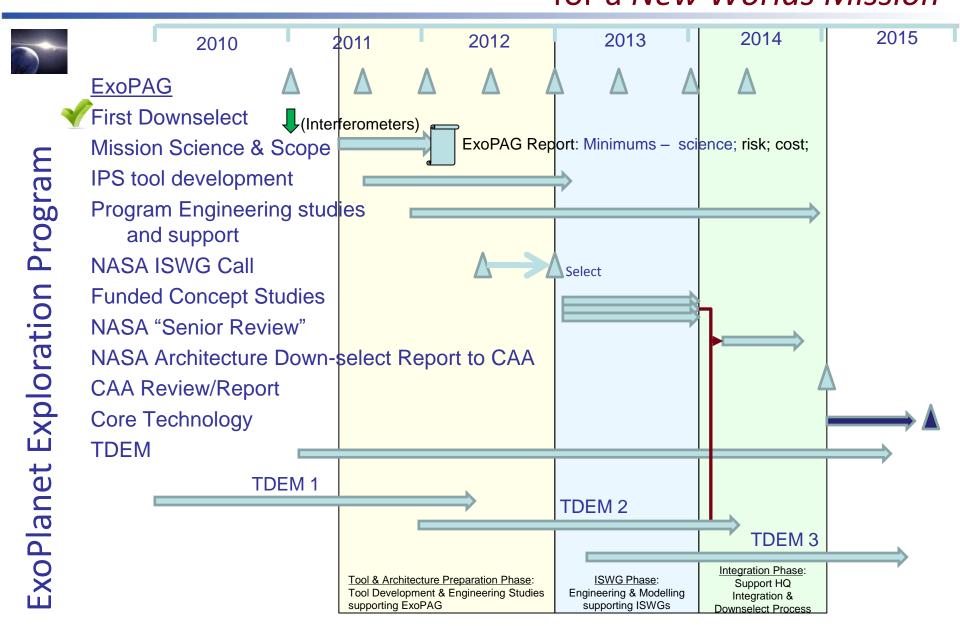
Anet program

NATIONAL RESEARCH COUNCIL
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# Concept Studies and Architecture Selection for a *New Worlds Mission*



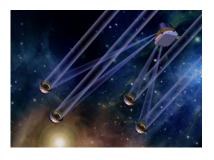




## Infrared Interferometer Concepts

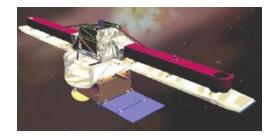


- Wavelength range: 3-8 microns, 6-18 microns
- Cryogenic optics, diffraction limited
- Primary mirrors (0.5–2-m diameter) of beryllium or silicon carbide
- Cryogenic deformable mirrors for wave-front correction
- Cryogenic delay-lines and fine-steering mirrors
- Mid-infrared single-mode fiber optics
- Cryocoolers for detectors
- Passive cooling of optics to 35–40 K
- Formation flying



**TPF-I** Terrestrial Planet Finder Interferometer

21 June 2011



**FKSI** Fourier-Kelvin Stellar Interferometer



Research slowed or halted Low-priority for NASA funding



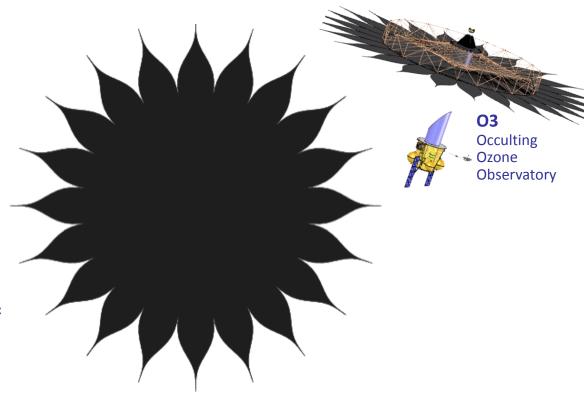


## **Starshade** Concepts



Starshade Exoplanet Telescope ŃWO Star **New Worlds Observer** 

- Nominally 4-m diffraction-limited primary mirror
- Mirror technology is not a driving concern
- Challenging technology is in the starshade itself and GN&C



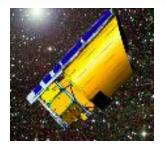


## **Coronagraphs**

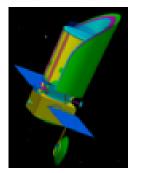




**TPF-C** Terrestrial Planet Finder Coronagraph



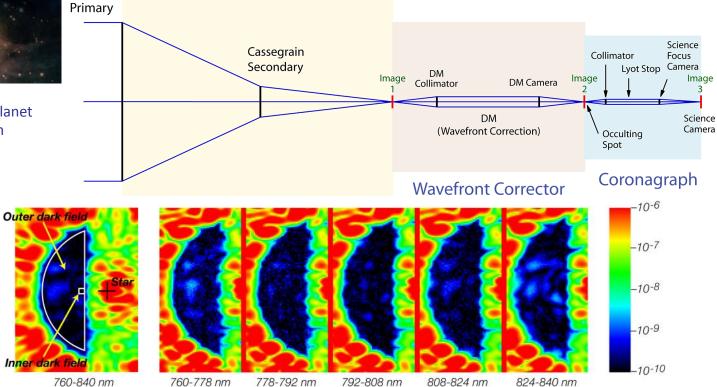
**PECO** Pupil-Mapping Exoplanet Coronagraphic Observe



**ACCESS** Actively-Corrected



Xinetics Inc (NGAS)



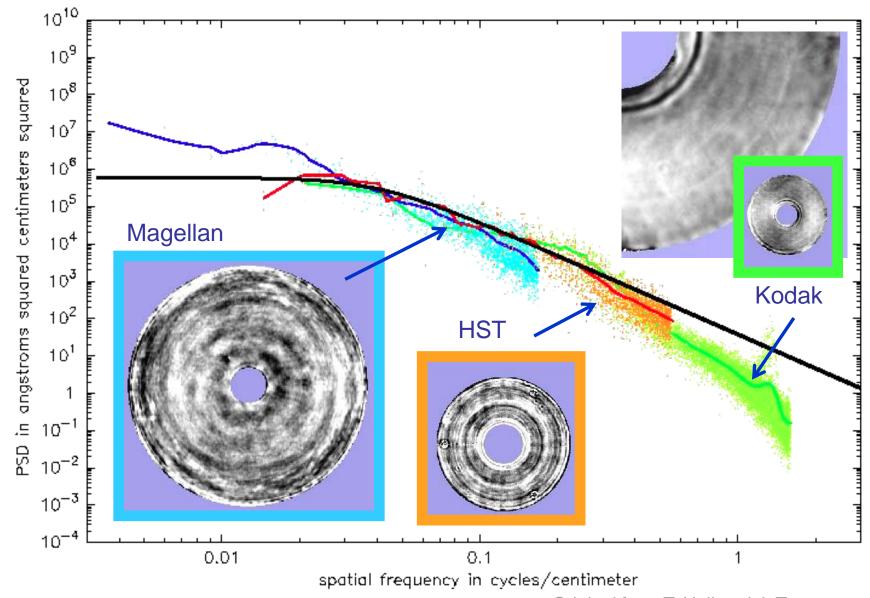




## **Primary Mirror: Surface Figure**







Original from T. Hull and J. Trauger



## **Technology Demonstrator Mirror Requirements**



### **TDM Baseline Concept**

1.9-m diameter, off-axis segment

• Surface errors < 5 nm rms in mid-spatial frequencies

Areal density of 47.5 kg/m<sup>2</sup> (60 kg/m<sup>2</sup> required)

Advanced waterjet lightweighted

ULE glass

On orbit PV quilting < 1mn</li>

Parameter	Range	aduii	Goal	
Parameter Range Goal SURFACE ERROR REQUIREMENTS				
Low Spatial Frequency	< 0. 10	10 nm rms	5 nm rms	
Mid Spatial Frequency	0.025 cycles/cm	5 nm rms	2.5 nm rms	
High Spatial Frequency	0.5-10 cycles/cm	1.4 nm rms	0.7 nm rms	
Micro-roughness	> 10 cycles/cm	10 Å rms	5 Å rms	
COATING RESIDUAL REFLECTANCE REQUIREMENTS				
Mid Spatial Frequency	0.025-0.5 cycles/cm	< 0.3% rms	≤ 0.1 % rms	

The revised HST-like surface requirements for a TPF-C are described in the following paper: S. B. Shaklan, J. J. Green, D. M. Palacios, "The Terrestrial Planet Finder Coronagraph Optical Surface Requirements," Proc. SPIE 6265, 62651L (2006).



## Major Engineering Requirements for TPF-C



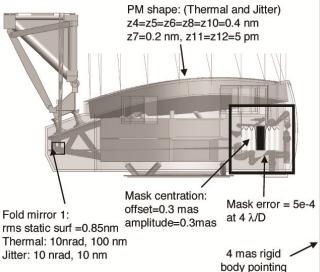
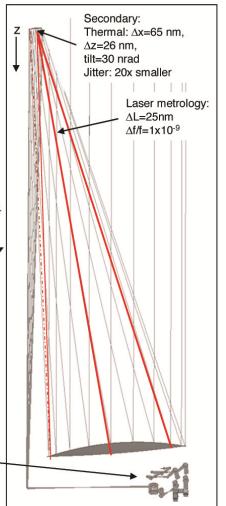


Figure 5. We identify the major engineering requirements to meet the dynamic error budget. Thermally induced translations lead to beam walk that is partially compensated by the secondary mirror. Jitter is partially compensated by the fine guiding mirror.

Coronagraph optics motion: Thermal:10nrad, 100nm \_\_ Jitter: 10 nrad, 10 nm



Laser metrology required for sensing and control

Key requirements
1–2 orders of
magnitude tighter
for a 3.8-m telescope
operating at 2λ/D

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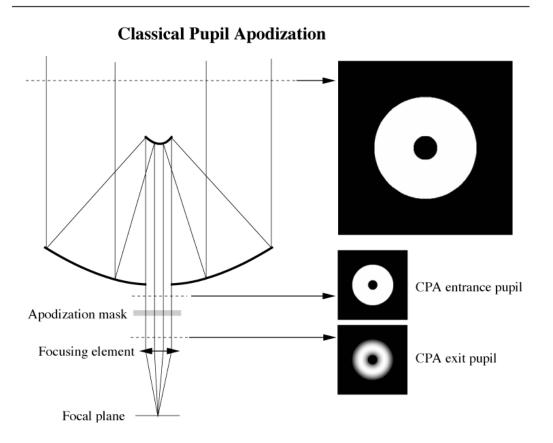
- S. B. Shaklan, L. Marchan, J. J. Green, O. P. Lay, "Terrestrial Planet Finder Coronagraph Dynamics Error Budget," Proc. SPIE 5905 (2005).
- S. B. Shaklan, L. F. Marchan, J. E. Krist, M. Rud, "Stability error budget for an aggressive coronagraph on a 3.8-m telescope," Proc. SPIE 8151, San Diego, August 2011.

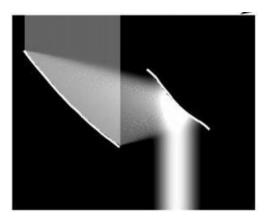


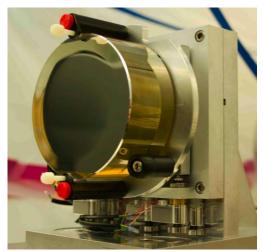
# NASA

## Phase-Induced Amplitude Apodization









14

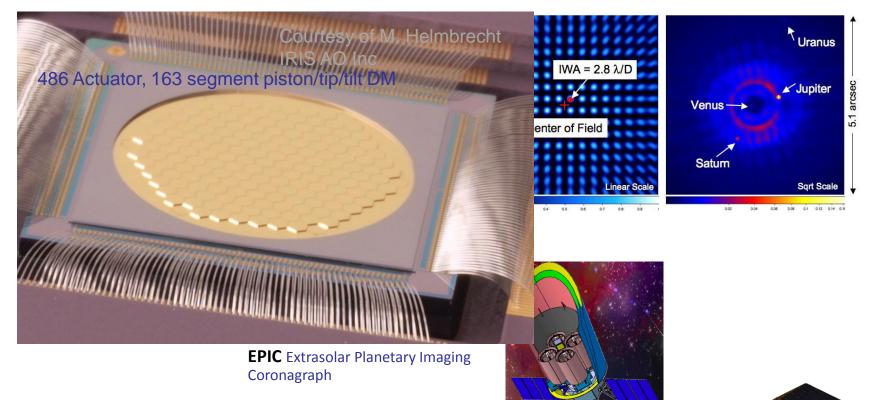
Tinsley Laboratories

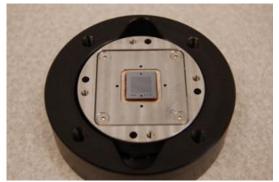
O. Guyon, S. Shaklan, M. Levine, et al. "The Pupil mapping Exoplanet Coronagraphic Observer," Proc. SPIE 7731, 77319 (2010)



## Visible Nulling Coronagraphs







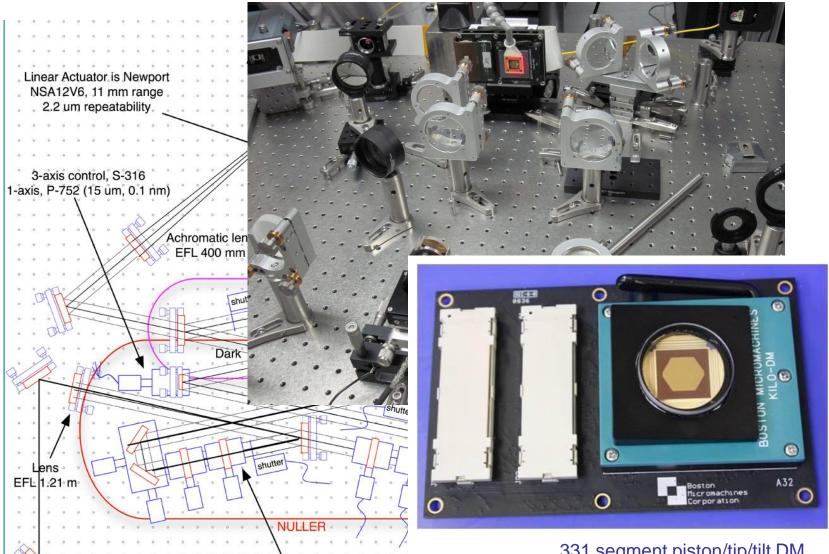


**ATLAST** Advanced Technology Large-Aperture Space Telescope



## **APEP Testbed at JPL**





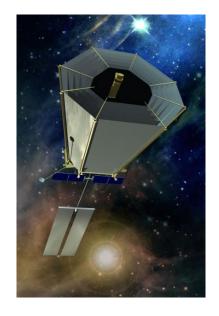
331 segment piston/tip/tilt DM Courtesy of T. Bifano, Boston Micromachines

Translation



## Gallery of Exoplanet Mission Concepts

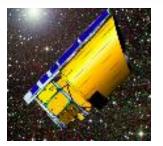




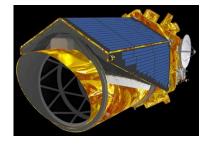
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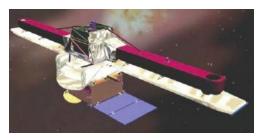
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Large-Aperture Space Telescope

## Coronagraph

- Starlight entering the telescope must be suppressed
- Diffraction artifacts must be controlled or mitigated
  - Off-axis un-obscured aperture design
  - Coronagraphic masks
  - Deformable mirrors
- Visible Nulling Coronagraph
  - Similar constraints as above
  - Instrument is an interferometer
  - Well suited for use with segmented-mirror telescopes, yet with extremely challenging requirements

### Starshade

- Starlight never enters the telescope no scattered light to suppress
- Diffraction effects are controlled by the shape of the starshade
  - The telescope must be diffraction limited
- Mirror technology is not a driving concern



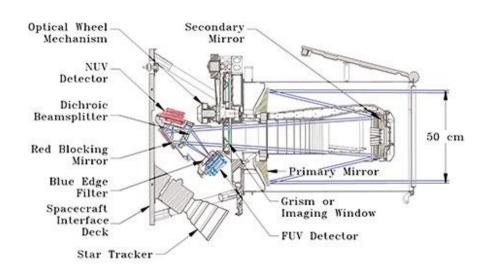


## UV/Optical and Exoplanet Mission?



## Example: GALEX (2003–present)





The most cost-effective way to improve sensitivity for a UV telescope would be to develop better UV detectors and more efficient coatings – not by going to larger mirrors

Exoplanet missions need mirrors at least 4-m in diameter

A starshade design would not impose additional constraints on a UV telescope, other than adding navigator beacons or sensors

A coronagraph design typically requires an unobscured, off-axis primary, and polarization-preserving multi-layer coatings, thus imposing numerous additional requirements on the telescope





- There are several possible approaches to designing exoplanet missions
  - Coronagraphs
  - Interferometers
  - Starshades
- Wavefront sensing and control is the central concern, not mirror size
  - Starlight suppression with deformable mirrors
  - Thermal and structural stability
  - Metrology for sensing and control
- Diffraction-limited 4-m class (or larger) primary mirror is needed to detect Earth-like planets
  - Surface figure similar to HST required
  - Stability tolerances of coronagraphs are achievable when larger primaries are used in conjunction with 8<sup>th</sup>-order masks
  - Smaller primary mirrors could be used with aggressive coronagraph designs, but the stability tolerances are then extremely challenging
- Long term vision for large telescope development includes space-based segmented-mirror telescopes using actively-controlled glass segments or silicon carbide hybrid-mirror designs



## Acknowledgements



This work was conducted at the Jet Propulsion Laboratory, California Institute of Technology, under contract with the National Aeronautics and Space Administration

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# Backup

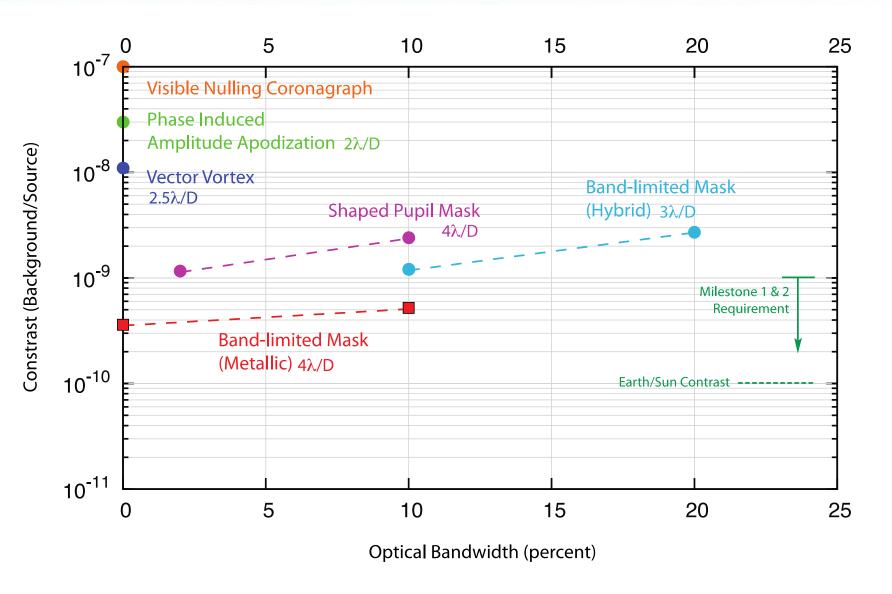


## State-of-the-Art in Coronagraph Laboratory Experiments





**ExoPlanet Exploration Program** 



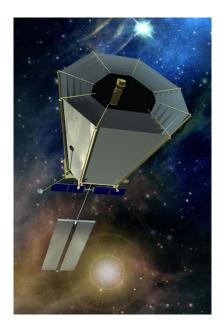




# Coronagraph Low-order Aberration Sensitivity For $3.5 \times 8.0$ -m Primary Mirror at $4\lambda/D$



Allowed Root Mean Square Wave Front for Contrast =  $1 \times 10^{-12}$  at  $4\lambda/D$ 3.5 x 8.0-m elliptical primary mirror



TPF-Coronagraph

Aberration	Eight order mask	Fourth order mask	Relaxation Ratio
Tilt	3.764	0.236	16
Focus	0.333	0.003	132
Astigmatism	0.464	0.004	132
Coma	0.088	0.001	185
Trefoil	0.201	0.002	132
Spherical	0.003	4.0E-4	7.2

24

Note. – Aberration units are in nm rms

S. B. Shaklan and J. J. Green "Low-order aberration sensitivity of eighth-order coronagraph masks," Astrophysical Journal 628, 474-477 (2005), Table 1 rescaled for a wavelength of 550 nm.



# NASA

## 2010 Phase 1: S2 Science Instruments



#### **BEAM Engineering for Advanced Measurements**

686 Formosa Avenue Winter Park, FL 32789-4523 Nelson Tabirian (407) 629-1282 10-1-S2.02-8374 JPL

Achromatic Vector Vortex Waveplates for Coronagraphy

### **Boston Micromachines Corporation**

30 Spinelli Place Cambridge, MA 02138-1070 Paul Bierden (617) 868-4178 10-1-S2.02-8461 JPL

Enhanced Reliability MEMS Deformable Mirrors for Space Imaging Applications

### Iris AO, Inc.

2680 Bancroft Way
Berkeley, CA 94704-1717
Michael Helmbrecht (510) 849-2375
10-1-S2.02-9446 GSFC
Picometer-Resolution MEMS Segmented DM

### **Bridger Photonics Inc.**

112 East Lincoln Bozeman, MT 59715-6504 Randy Reibel (406) 585-2774 10-1-S2.03-8200 JPL

Multi-Point Trilateration: A New Approach for Distributed Metrology

#### Vanguard Composites Group, Inc.

9431 Dowdy Drive San Diego, CA 92126-4336 Steven Sherman (858) 587-4210 10-1-S2.03-8698 JPL Carbon Fiber Reinforced, Zero CME Composites

#### L'Garde, Inc.

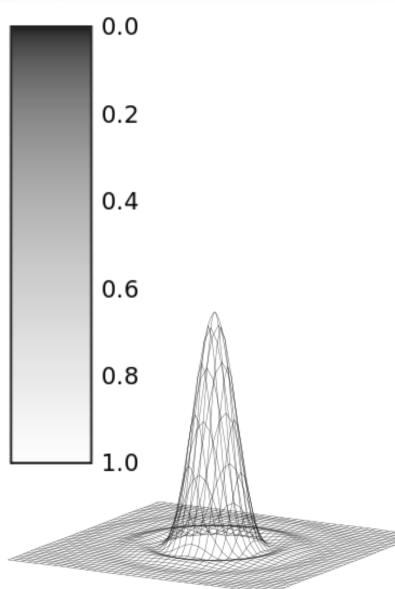
15181 Woodlawn Avenue Tustin, CA 92780-6487 Roger Garrett (714) 259-0771 10-1-S2.03-9177 JPL Thermally-Stable High Strain Deployable Structures

#### **Trex Enterprises Corporation**

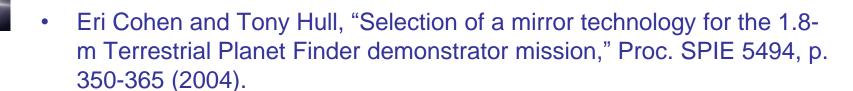
10455 Pacific Center Court San Diego, CA 92121-4339 Deborah Doyle (858) 997-9508 10-1-S2.04-9269 MSFC

Silicon Carbide Corrugated Mirrors for Space Telescopes

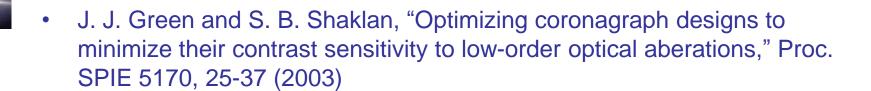








- D. A. Content et al. "Engineering trade studies for the Terrestrial Planet Finder Coronagraph primary mirror," Proc. SPIE 5867, 58670X (2005)
- S. B. Shaklan and J. J. Green, "Reflectivity and optical surface height requirements in a broadband coronagraph. 1. Contrast floor due to controllable spatial frequencies," Appl. Opt. 45, 5143-5153 (2006)
- K. Balasubramanian et al. "Low-cost high-precision PIAA optics for high contrast imaging with exo-planet coronagraphs," Proc. SPIE 7731, 77314U (2010)
- K. Balasubramanian et al. "Polarization compensating protective coatings for TPF-Coronagraph optics to control contrast degrading cross-polarization leakage," Proc. SPIE 5905, 59050H (2005)



- S. B. Shaklan and J. J. Green, "Low-order aberration sensitivity of eight-order coronographic masks," Astrophysical Journal 628, 474-477 (2005)
- S. B. Shaklan, et al. "The Terrestrial Planet Finder Coronagraph" Dynamics Error Budget," Proc. SPIE 5905, 59050D (2005)